

Lift Enhancement for Upper Surface Blowing (USB) Airplanes

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Outline

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 - **Active Flow Control (AFC)**
 - **AFC concept**
 - **Actuation modes**
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 - **Pulsed**
 - **Sprinkler**

Powered lift concepts

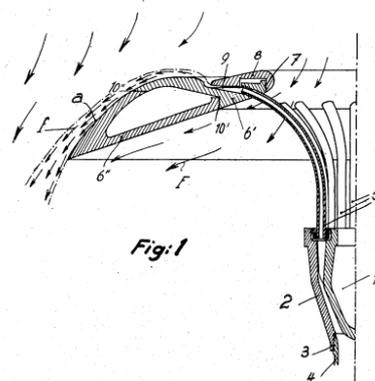
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- **Increase lift production by exploiting engine exhaust jet**
 - Externally Blown Flaps (EBF)
 - Internally Blown Flaps (IBF)
 - Upper Surface Blowing (USB)
- **Coanda Effect**

“Deviation of a plain jet of a fluid that penetrates another fluid in the vicinity of a convex wall”



Feb. 15, 1938. H. COANDA PROPELLING DEVICE 2,108,652
Filed Jan. 10, 1936 2 Sheets-Sheet 1

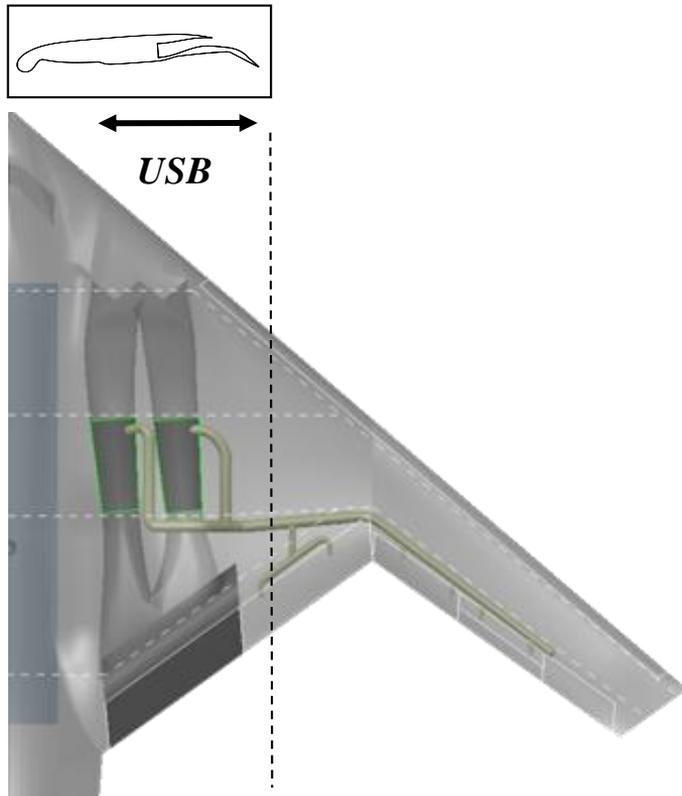


Experimental confirmation

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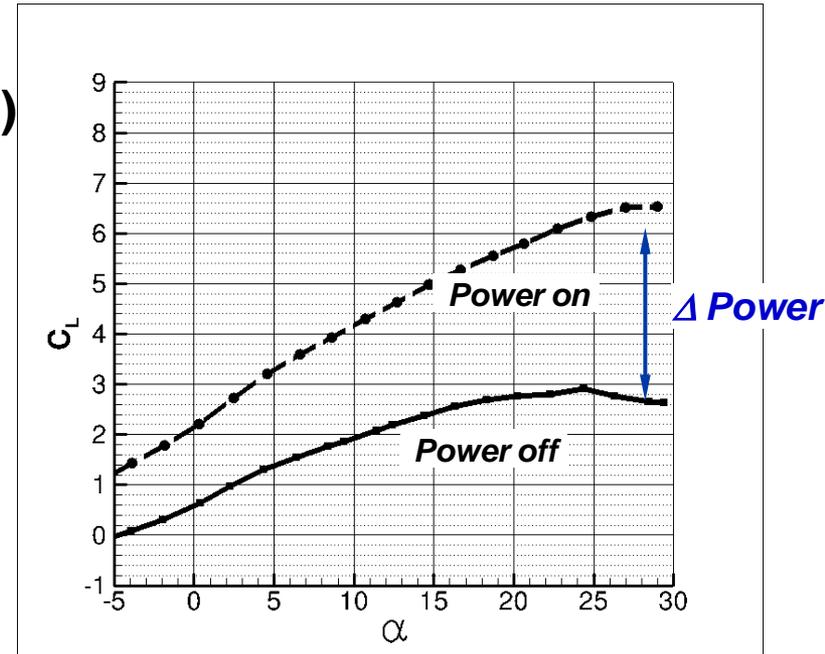
Speed Agile Concept Demonstrator

- AFRL
- Advanced Joint Air Combat System (AJACS)



Single element, simple hinge flap

3-element slotted slat and flap



Boeing V/STOL (Philadelphia, July 2007)

Current study

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- **Problem**

- **Premature jet peeling-off degrades lifting capability**

- **Ratio of Jet thickness to surface radius of curvature:**

- Thin jet

- Large radius of curvature

- **Edge effects**

- Spanwise extent of nozzle

- Edge vortex formation

- **Objective**

- **Develop technologies for enhanced USB systems**

- **Approach**

- **Geometrical modifications**

- **Active Flow Control**

Analysis approach

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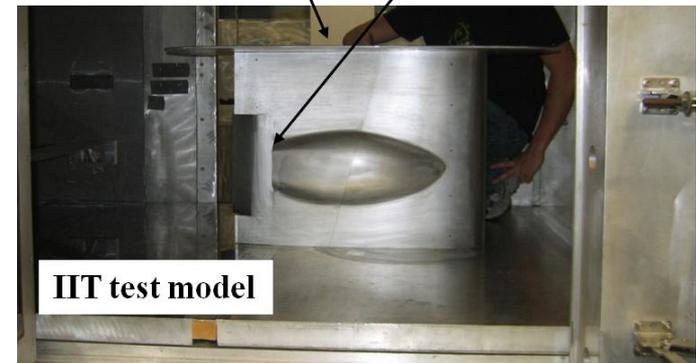
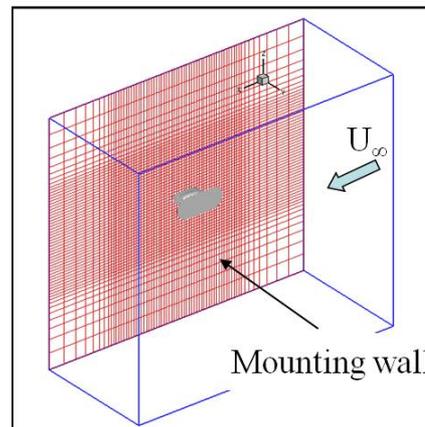
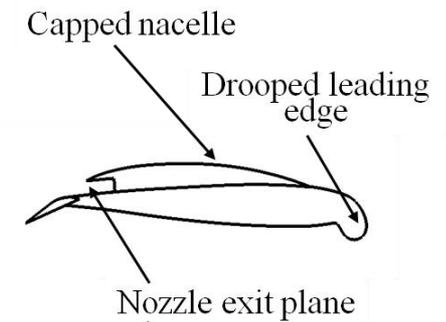
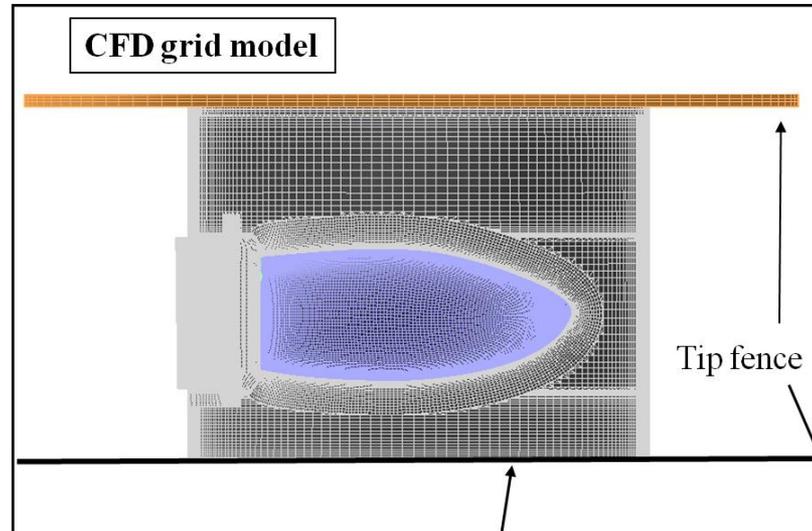
- **CFD based evaluation of the new technologies**
 - Validation
 - Application to USB enhanced concepts
- **Numerical tool**
 - Unsteady RANS (OVERFLOW with Boeing's modification)
 - Overset grid system
 - Upwind scheme
 - One-equation (S-A) turbulence model
 - Time-varying boundary conditions

Numerical validation

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Illinois Institute of technology 2007 test:

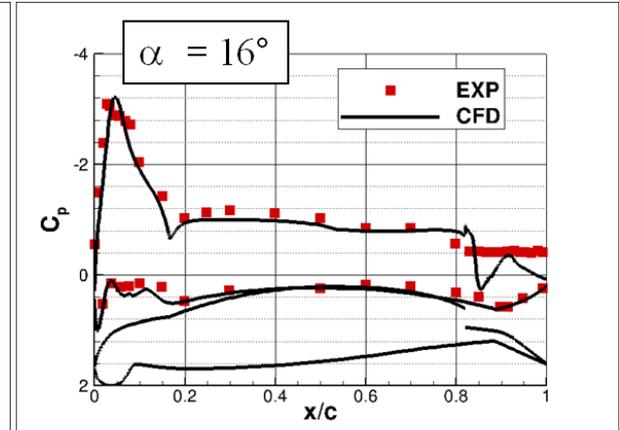
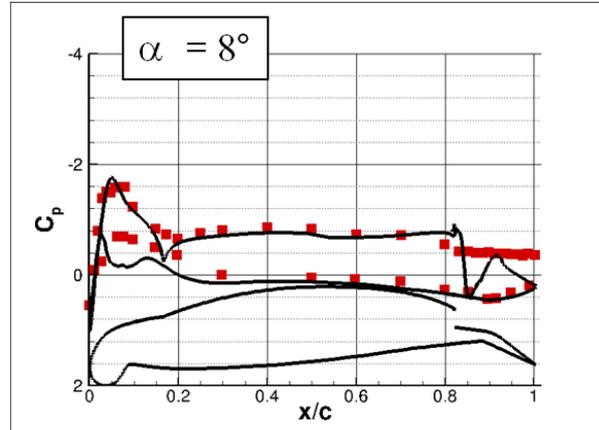
- CFD Model:
 - 10.5 million points
 - 34 zones



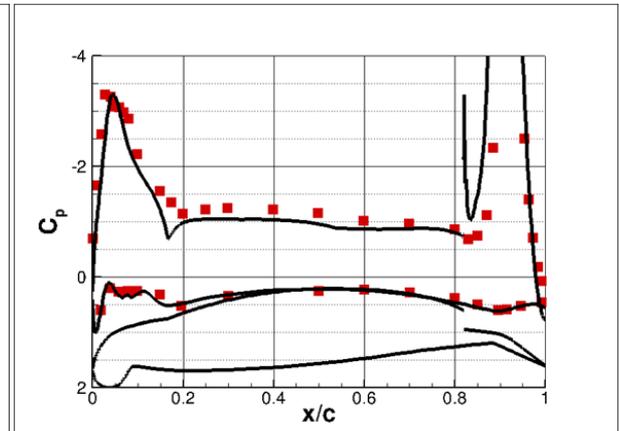
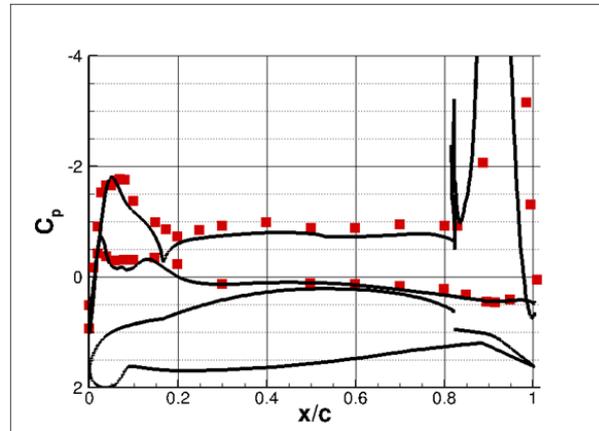
Numerical validation - sample

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Power OFF

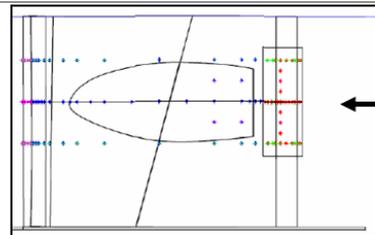


Power ON



$C_T = 0.8$

▪ near sonic nozzle



cut at centerline row

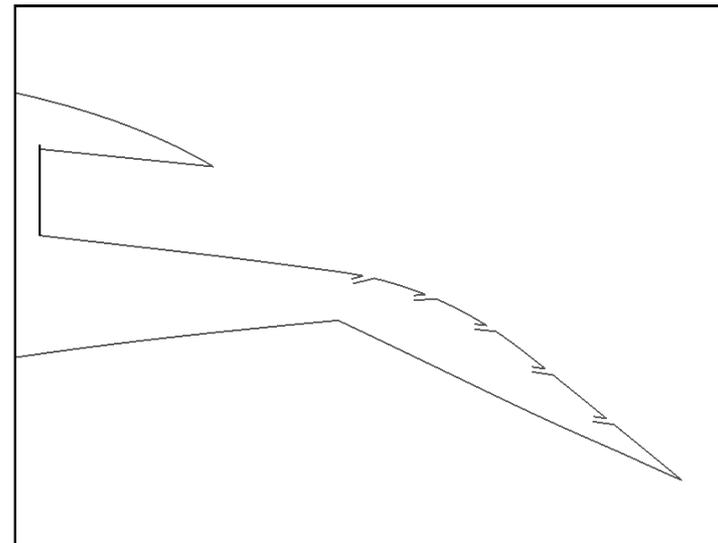
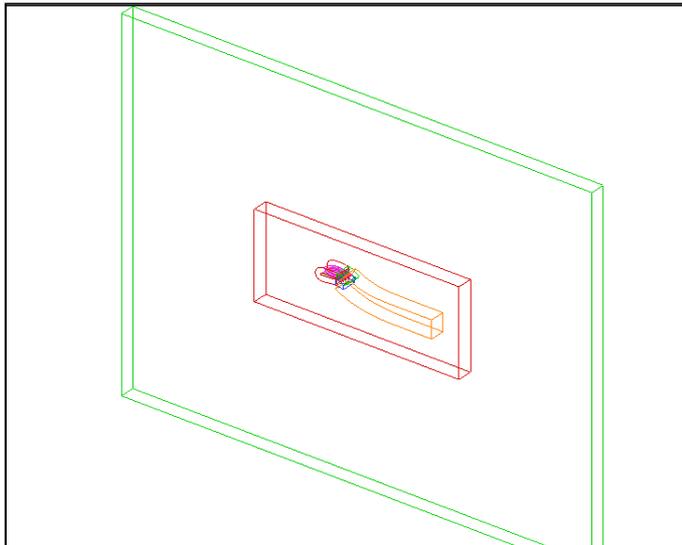
$U_\infty = 45 \text{ m/s}$
 $M_\infty = 0.132$

USB baseline configuration

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Computational test bed is based on the IIT model

- Wing of constant chord between 2 vertical walls
- Full span flap
- Nozzle aspect ratio of 6
- AFC slots embedded in flap upper surface
- Height of slot is 0.2% chord

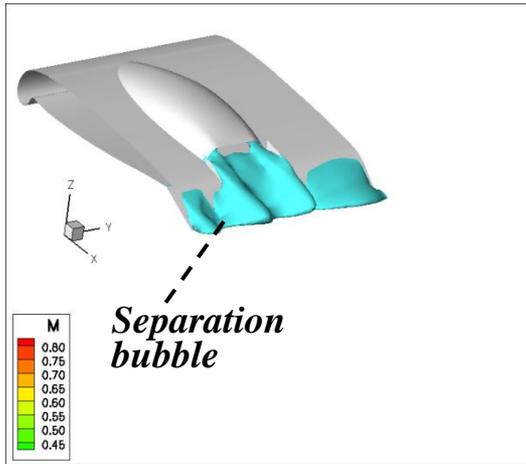


USB flow features

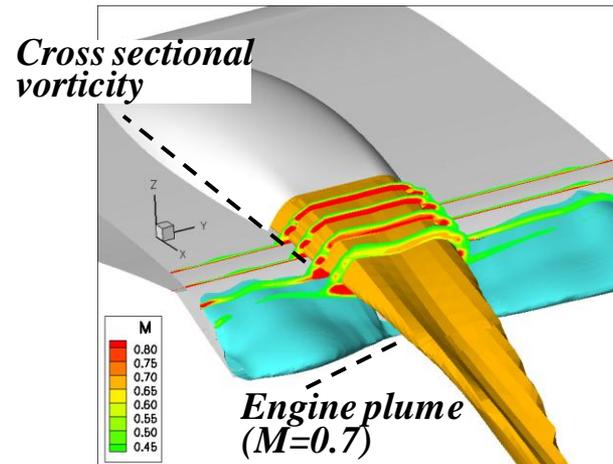
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Flow Features

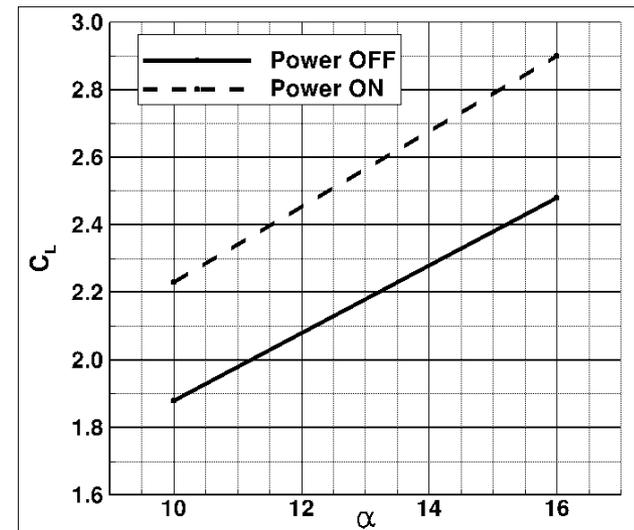
- Lift increment of 16%
- Flow reattachment behind engine
- Roll-up of engine plume at edge of nozzle



Power off



Power on (takeoff)



Techniques for enhanced USB

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Enhanced Coanda effect:

- Reshape flap upper surface
- Elongated nozzle

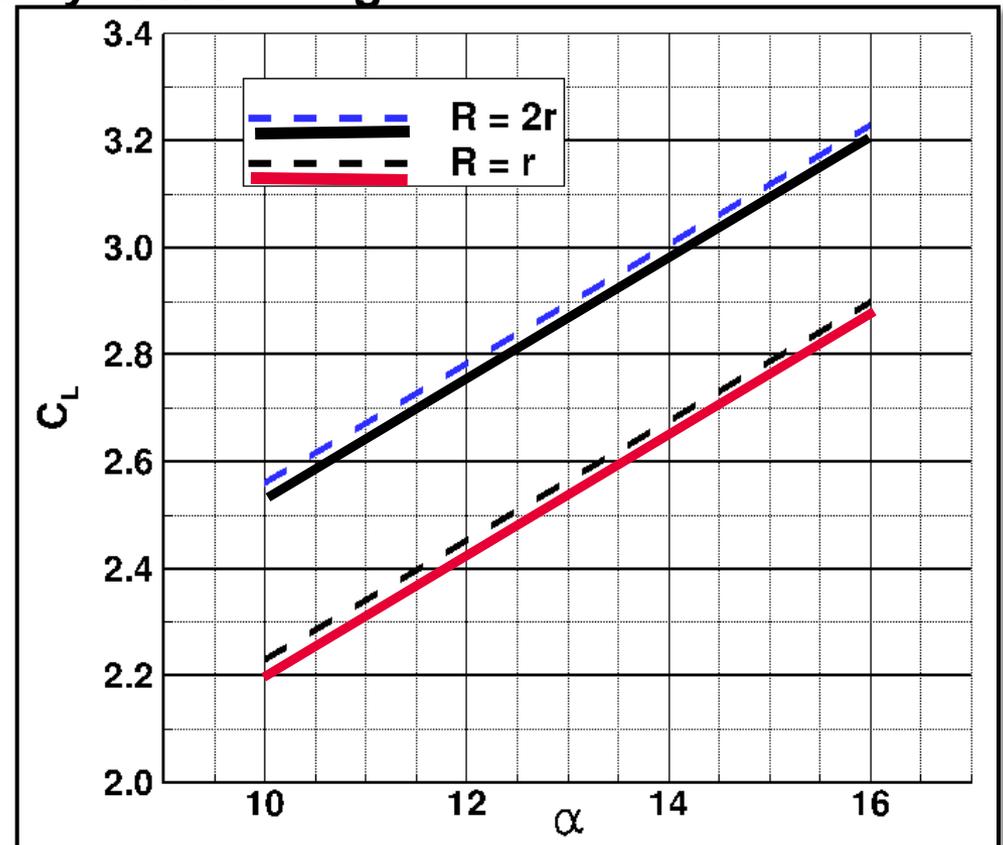
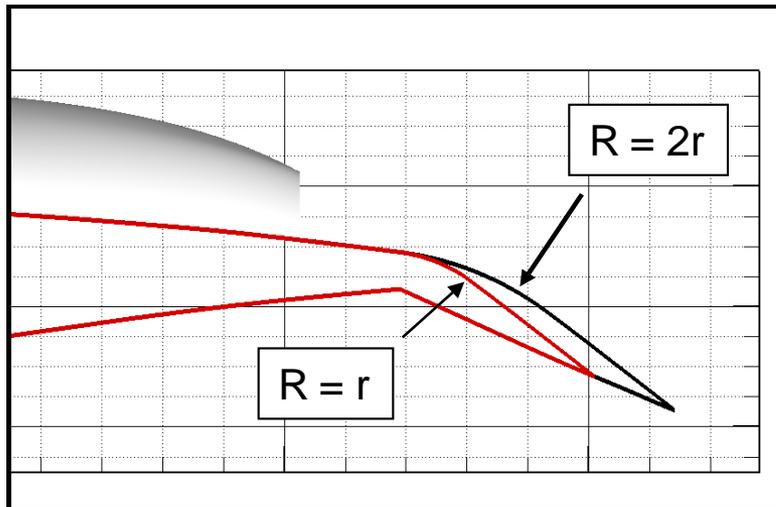
Mitigate nozzle edge effects:

- Nozzle edge effects
- Flow control methods:
 - Constant blowing
 - Pulse actuation
 - Sprinkler actuation

High curvature flap

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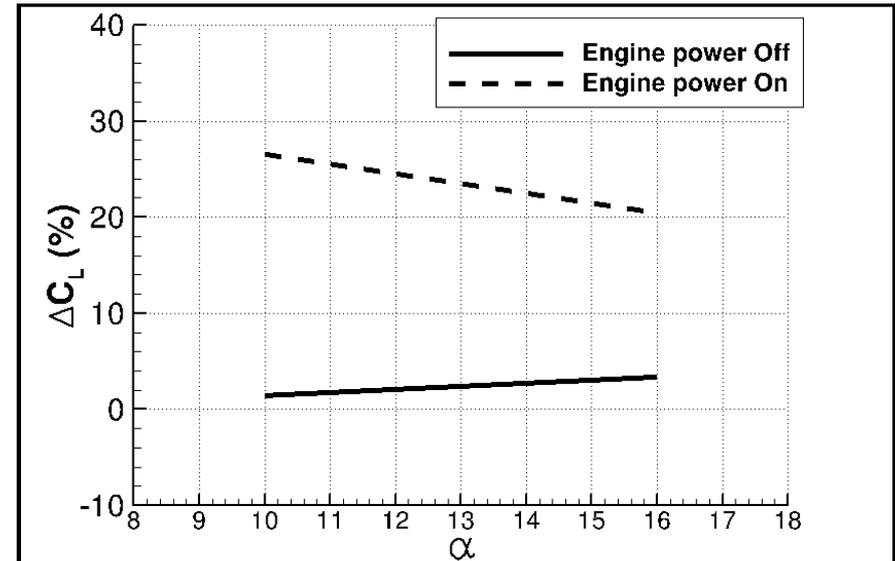
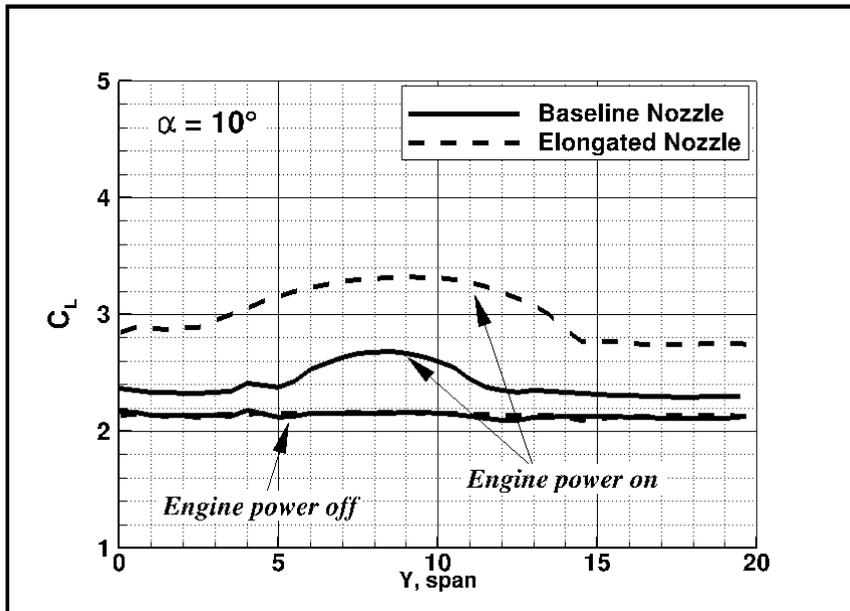
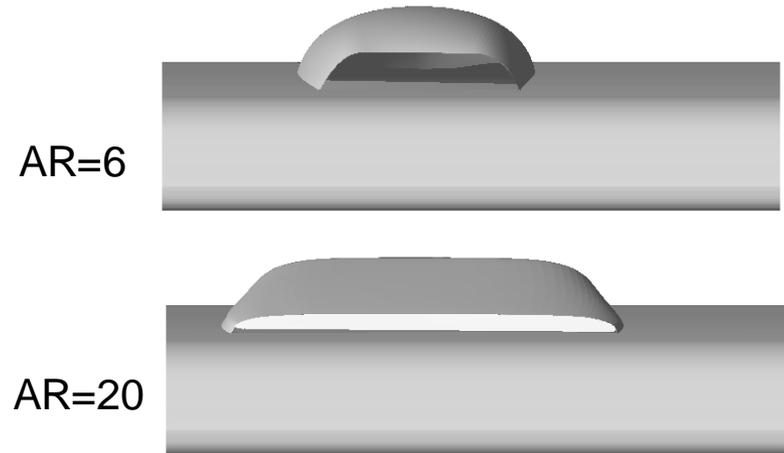
- 4.7% increase in wing chord
- Lift increase of 12%
- Augmented Coanda effect over only 28% of wing



High aspect ratio nozzle

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- Double nozzle width
- Fix nozzle exit area (fixed \dot{m})
- Reducing thickness of jet plume
- Larger USB affected area
- Flow is effectively more 2D
 - Smoother spanload distribution

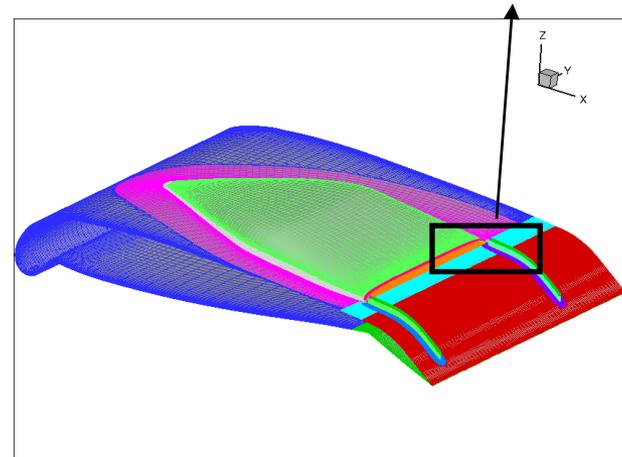
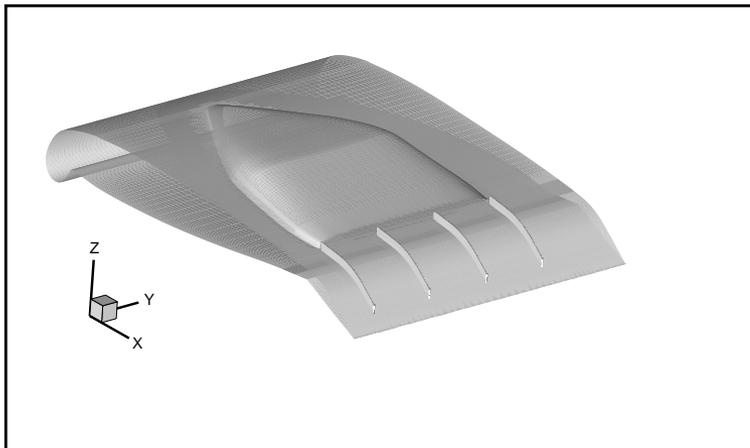
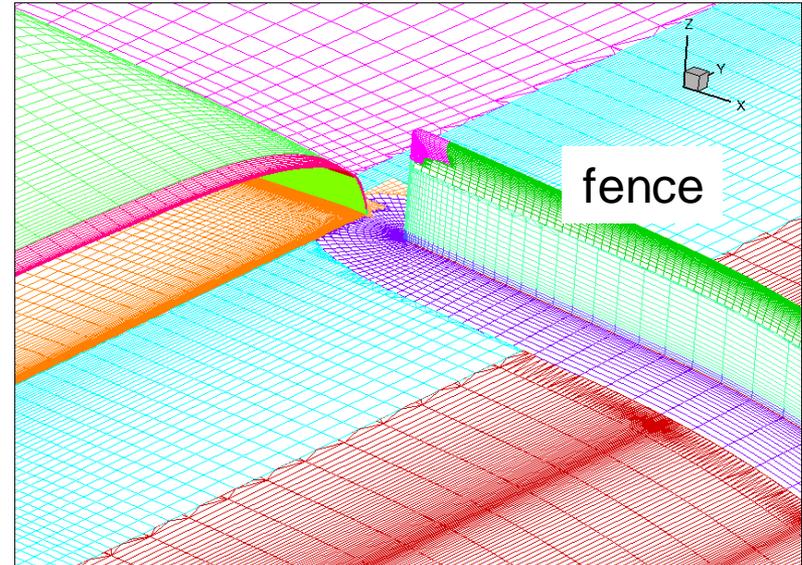


Flap fences

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- **Control edge effects**
- **Prevent vortex roll-up**
- **Reduce three-dimensionality of flow**

- **Variable height**
- **Fixed or automatically deployed**

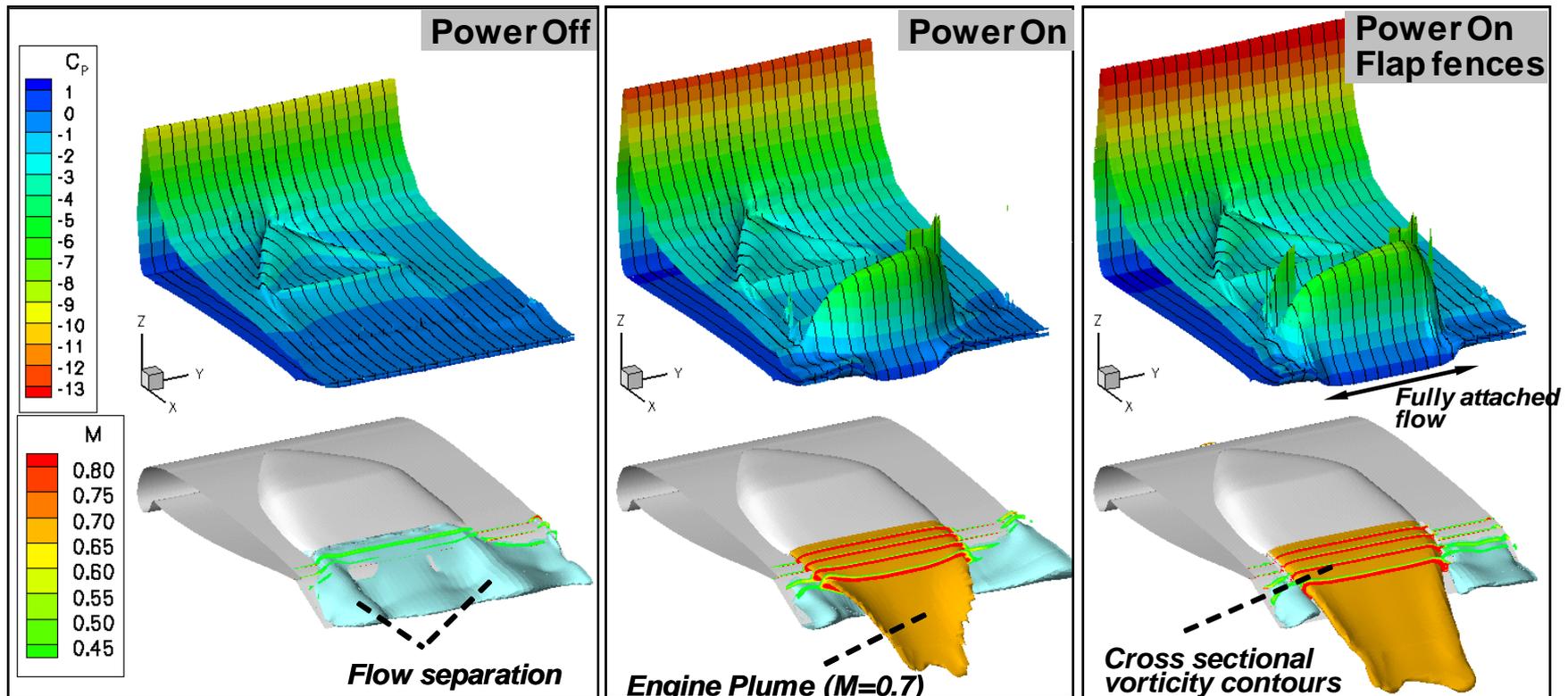


Flap fences effects

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- Wider plume
- Weaker edge vortices
- Fully attached flow over flap
- High LE pressure suction due to increased circulation

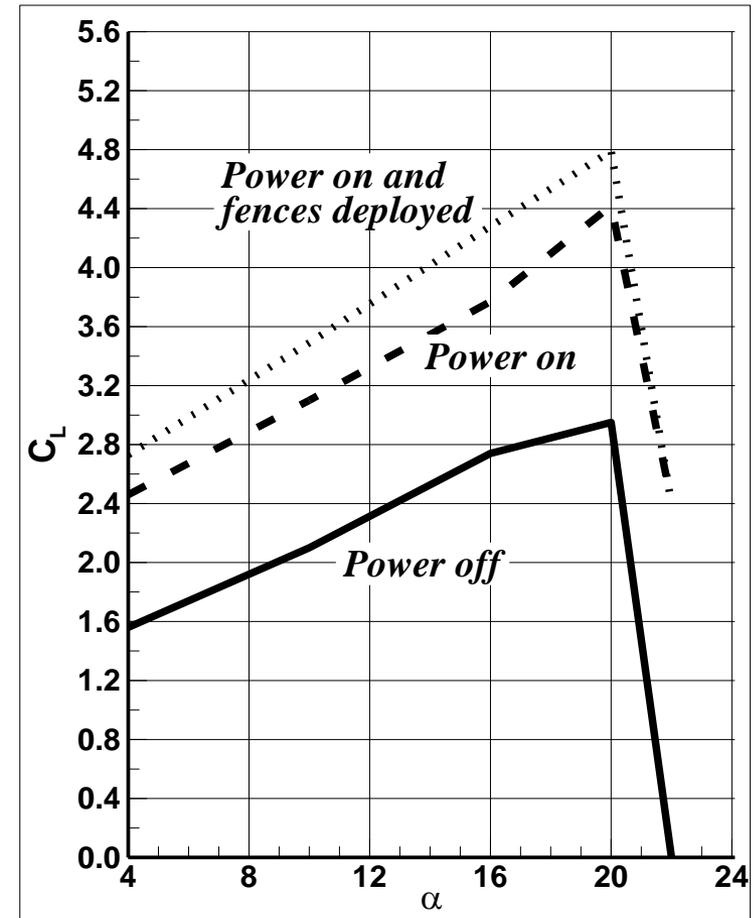
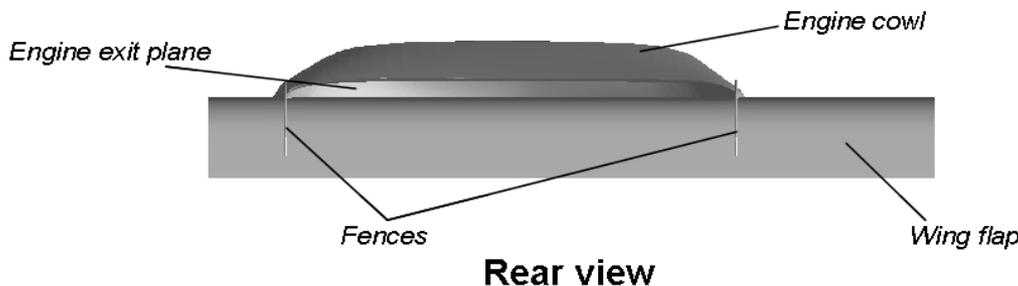
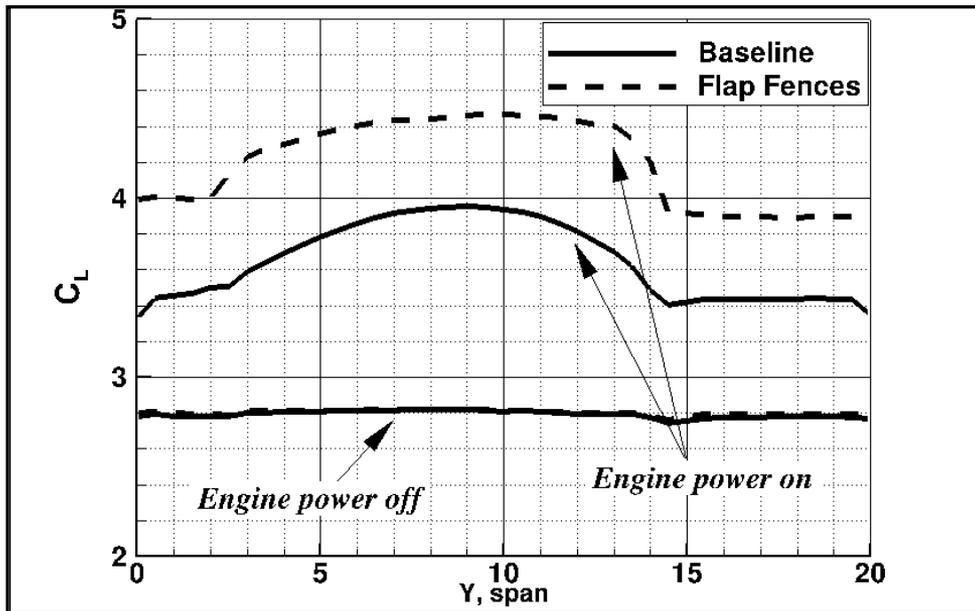
Elongated nozzle
 $\alpha = 16^\circ$



Flap fences performance

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- Higher load distribution is achieved across the entire span
- The flow becomes more “2d”-like between fences



Active Flow Control

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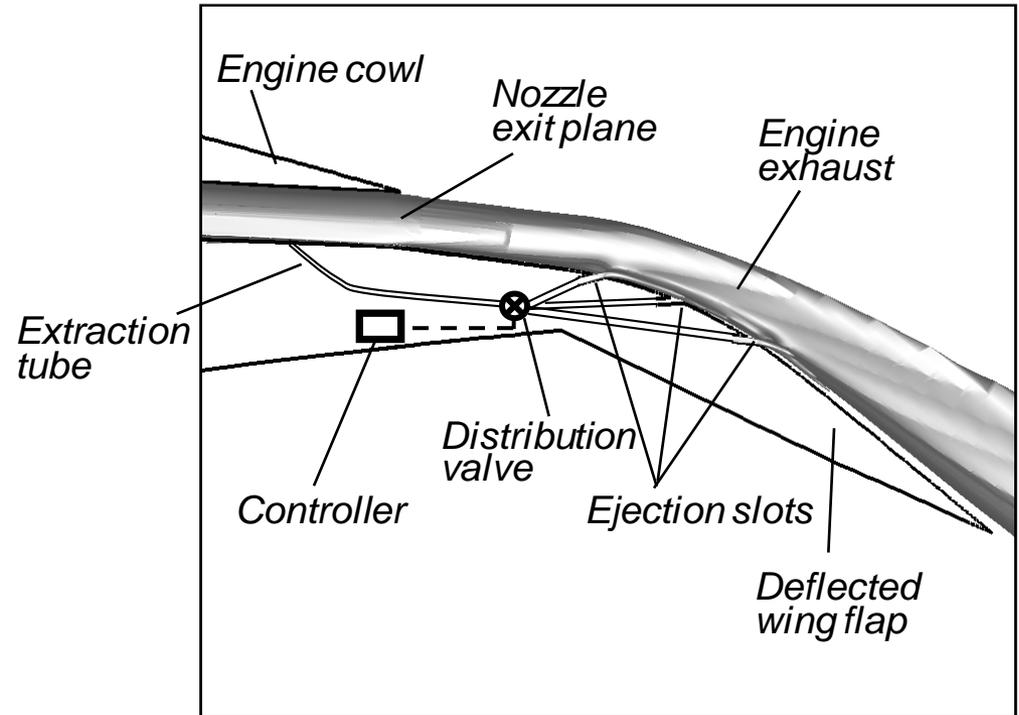
- **Lift enhancement through delayed flow separation at high angles of attack**
 - Zero mass flow devices
 - Fluidic device
 - Source of air
 - Engine bleed
 - Dedicated compressor
 - Actuation type
 - Constant blowing from a fixed orifice
 - Variable blowing
 - [Pulse actuation](#) ---periodic variation of jet velocity
 - [Sprinkler actuation](#) ---continuous swiveling of blowing jet

Active Flow Control setup

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Bleed off engine exhaust for actuation at plume's edges



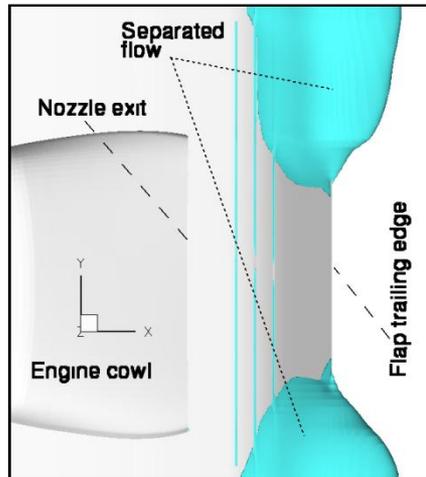
Cut through the engine

Constant blowing

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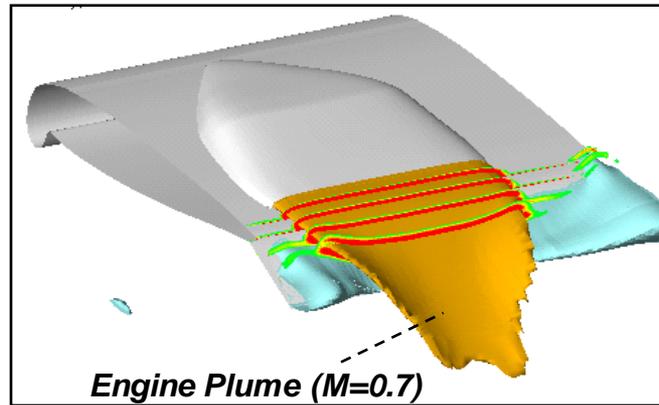
Actuation results in increased USB effectiveness

Top view

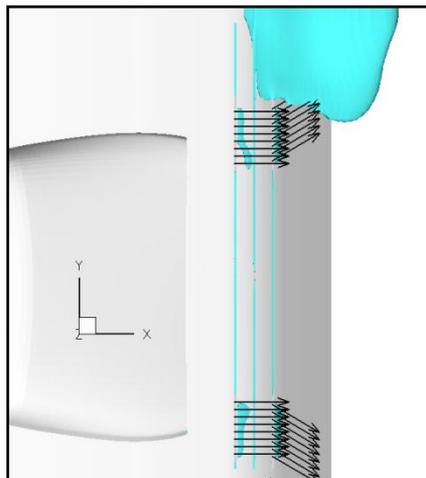
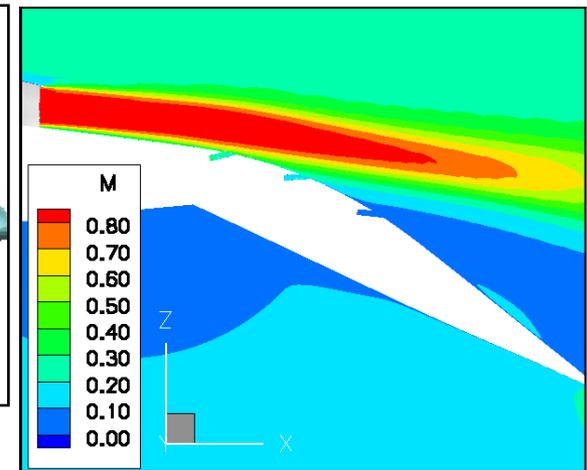


Control Off

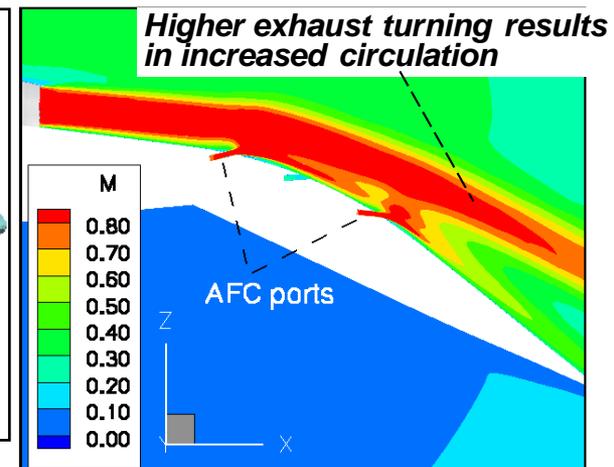
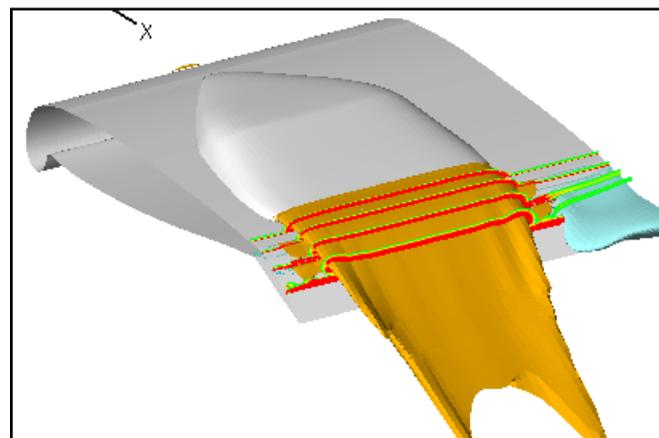
Exhaust plume



Vertical cut through ports



Control On

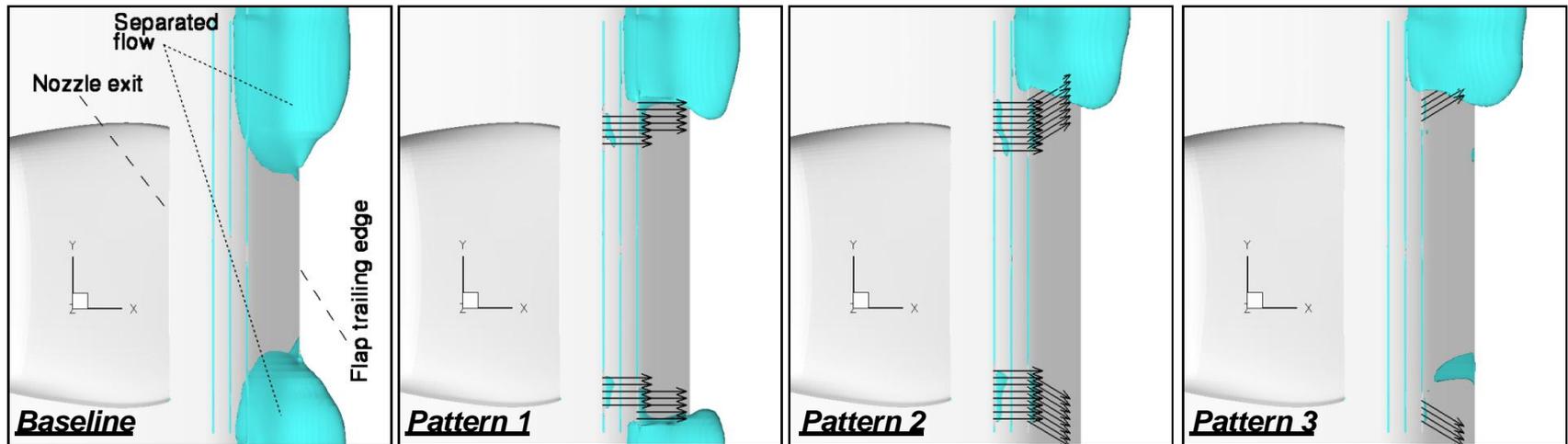


Constant blowing (cont.)

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Alternative actuation patterns:

- multiple flight conditions
- uncertainties in plume location

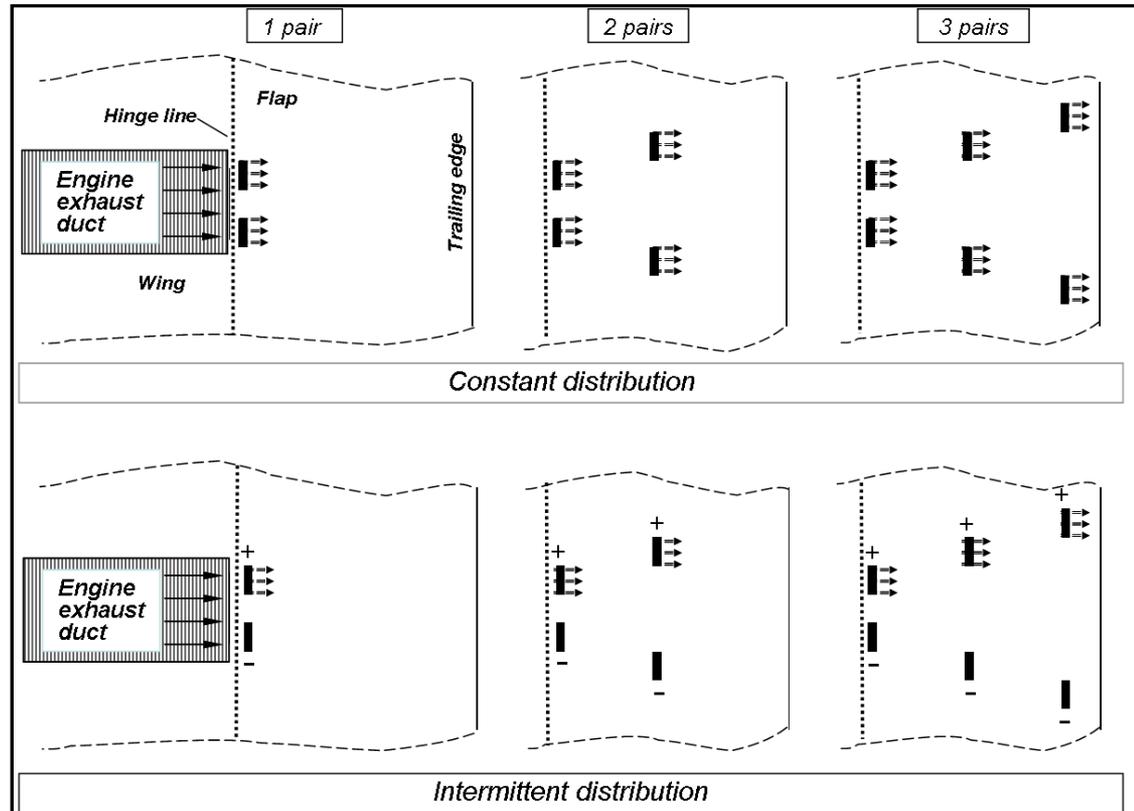


no actuation

Pulsed actuation

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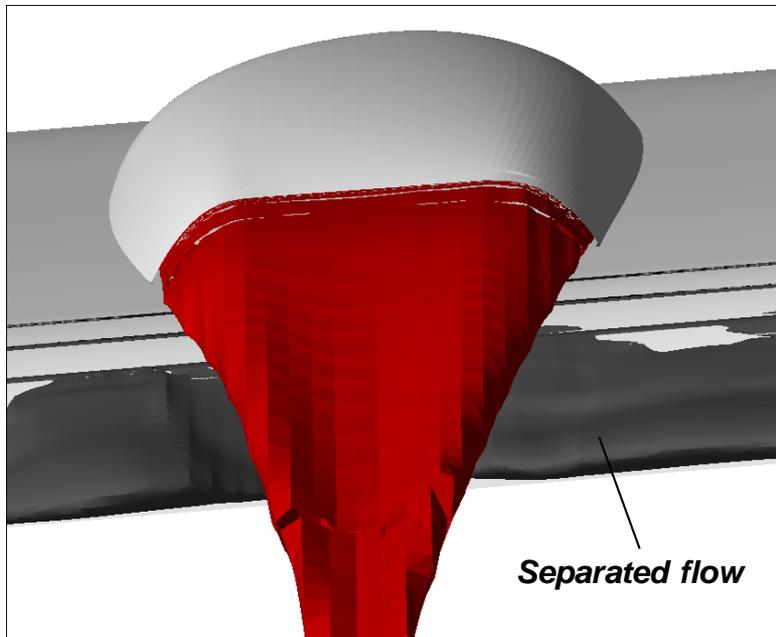
- **Reduce bleed air requirement**
 - Reduce engine size/weight
- **Wide range of actuations modes**
 - Physical distribution
 - Frequency
 - Phase



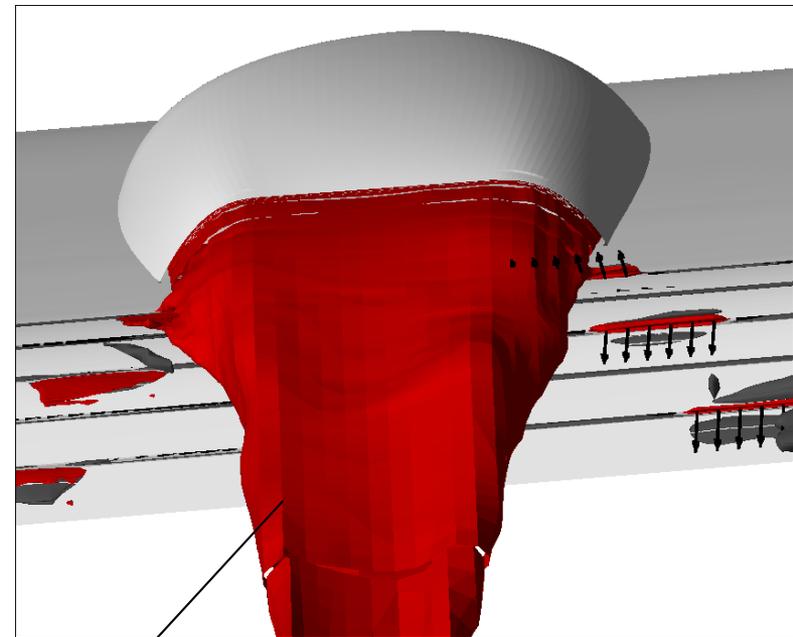
Pulsed actuation flow field

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Pulsed actuation cleans up flow separation



Baseline

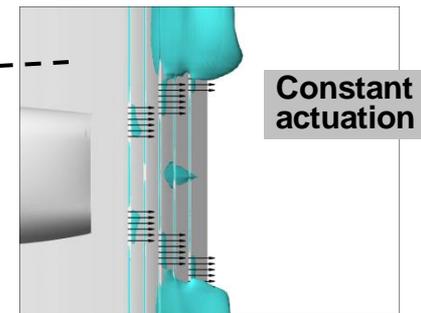
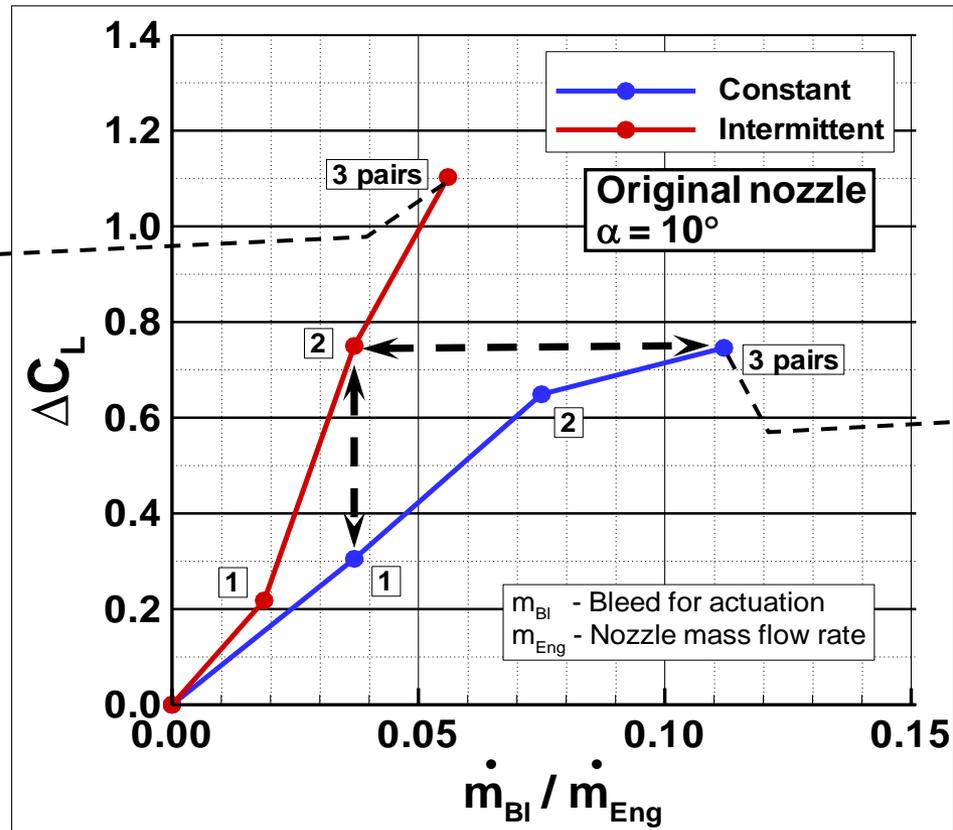
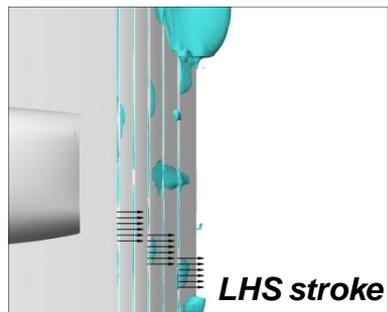
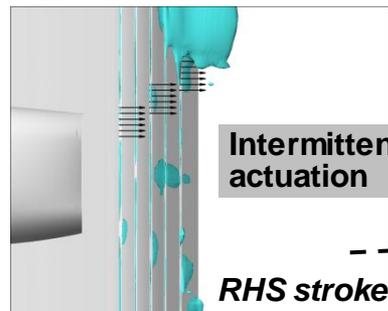


Intermittent (instantaneous)

Pulsed actuation effectiveness

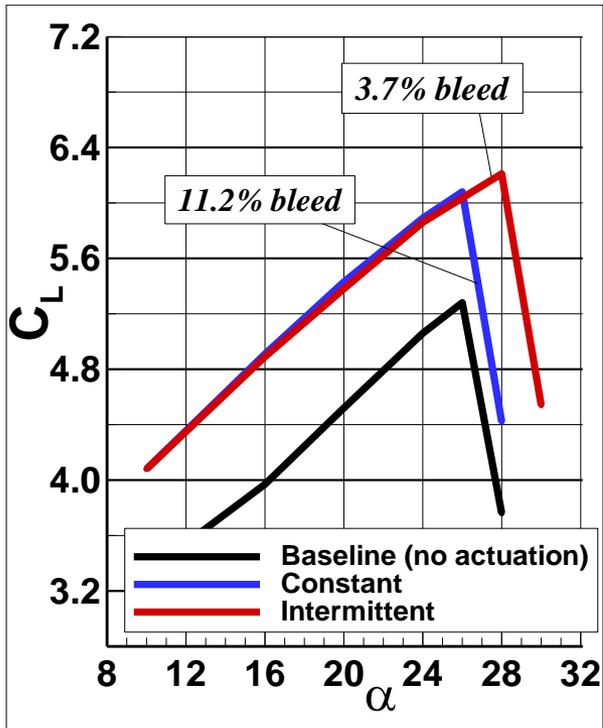
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Alternate between left and right blowing

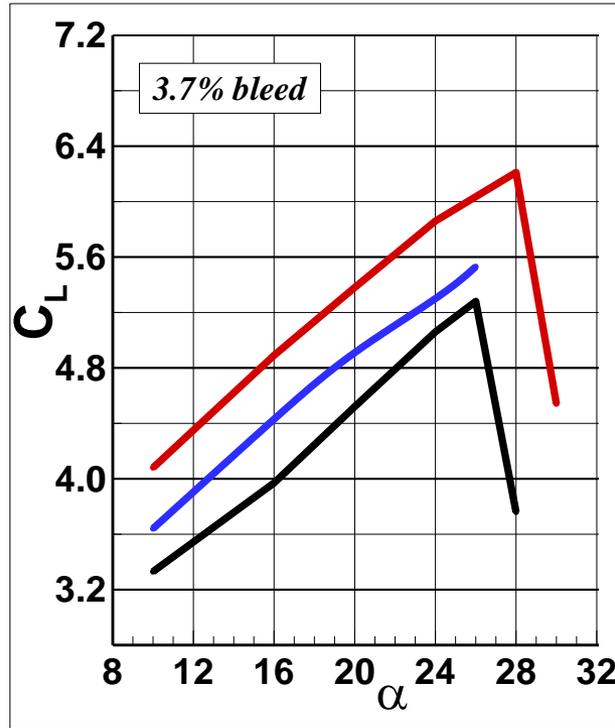


Advantages of pulsed actuation

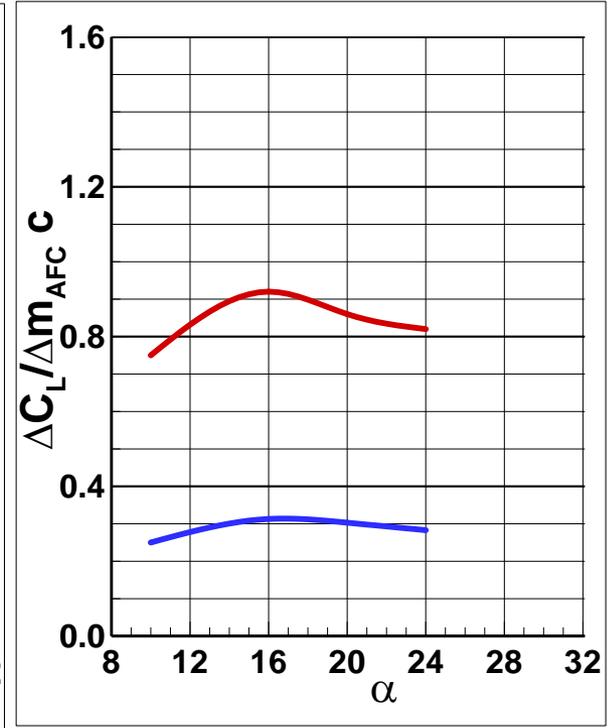
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Constant lift increment



Constant engine bleed

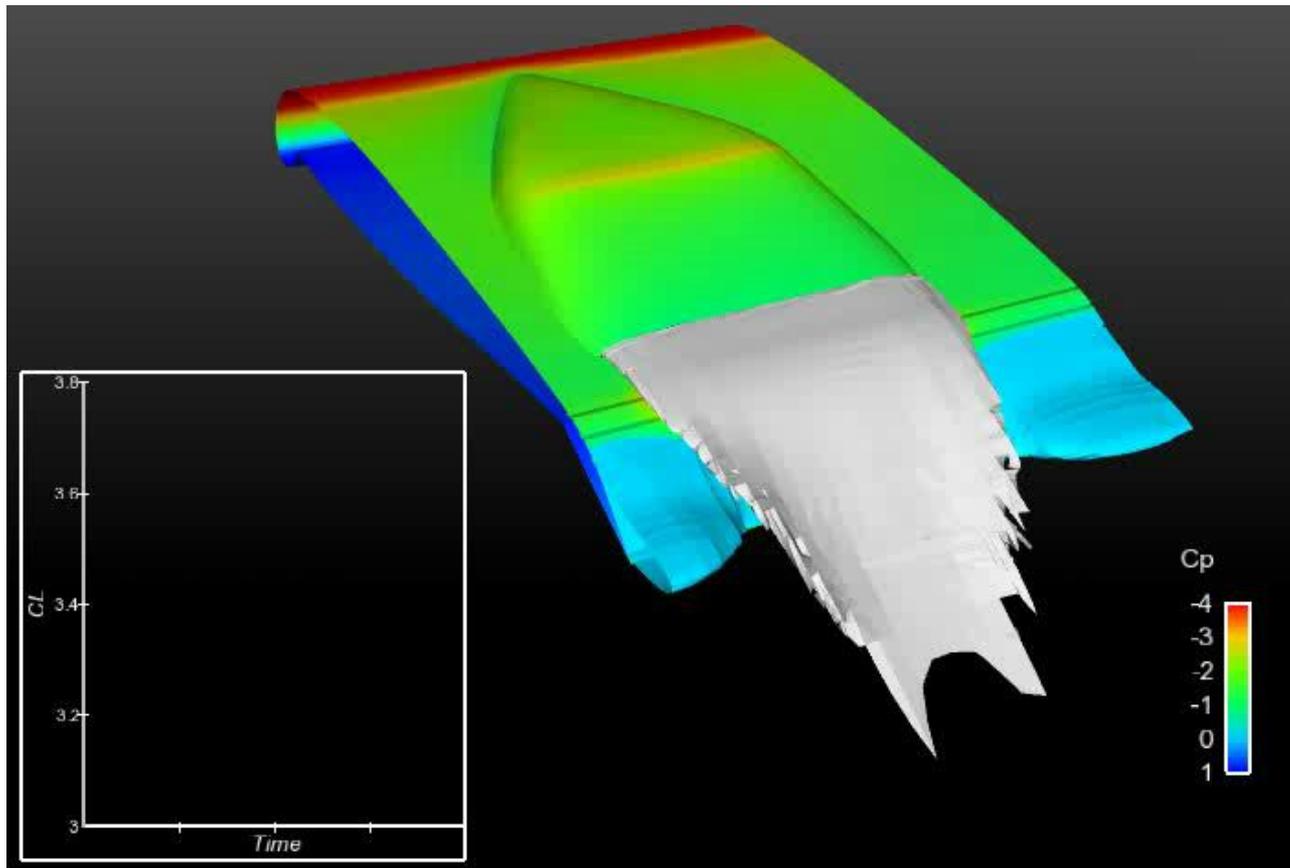


Efficiency

Pulsed actuation animation

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Elongated nozzle



Sprinkler Actuation

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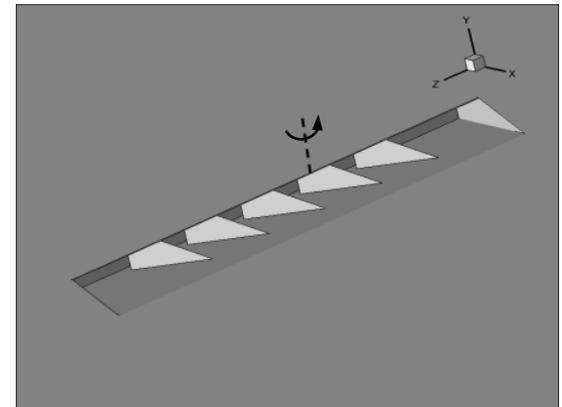
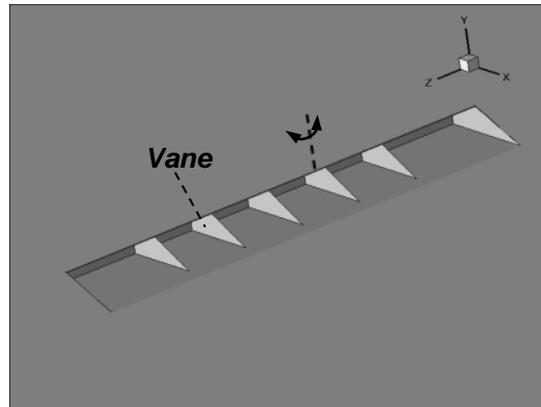
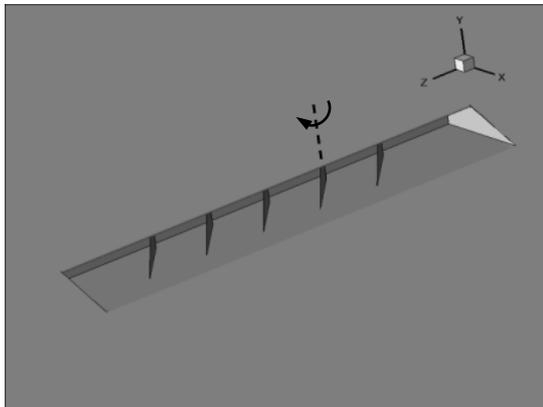
- **Multiple flight conditions requires flexibility:**

- Edge of jet plume might vary
- A-priori placement of ports may not be practical
- Need a “wide area coverage” device



Sprinkler actuator:

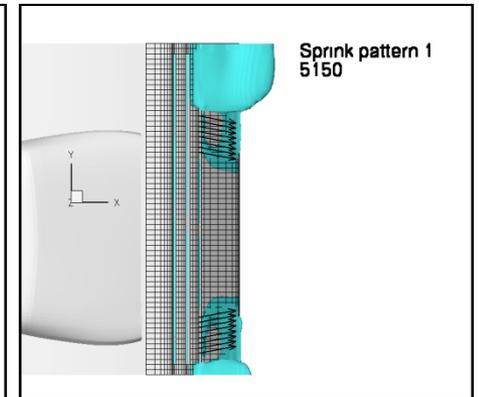
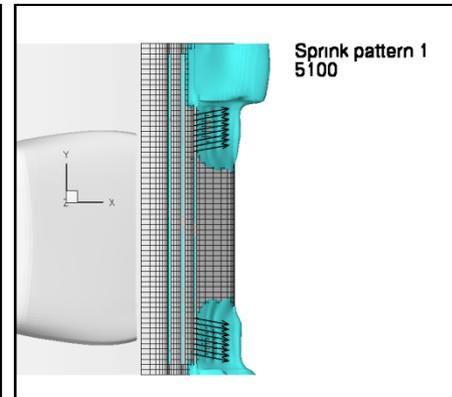
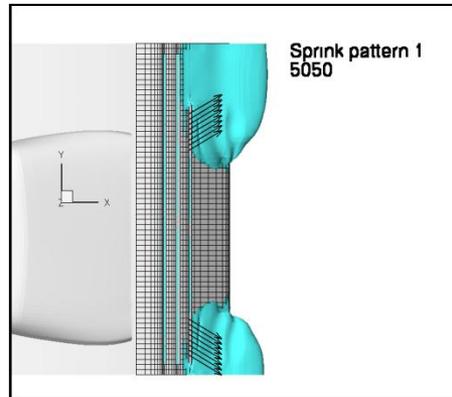
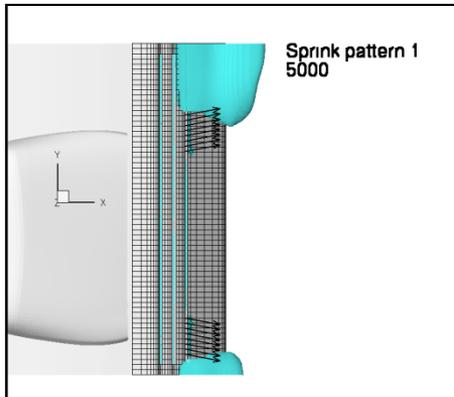
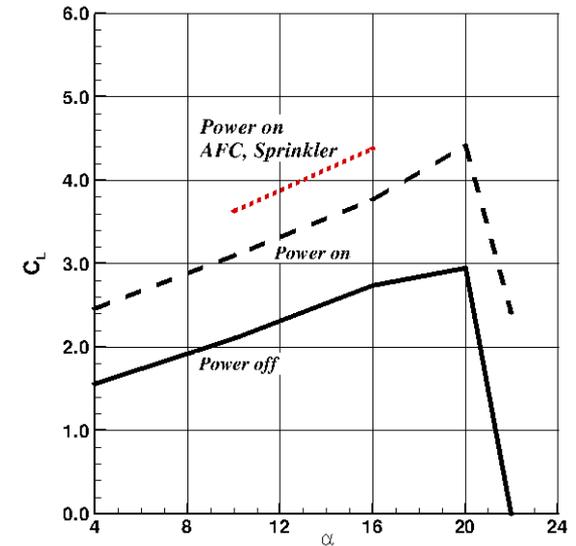
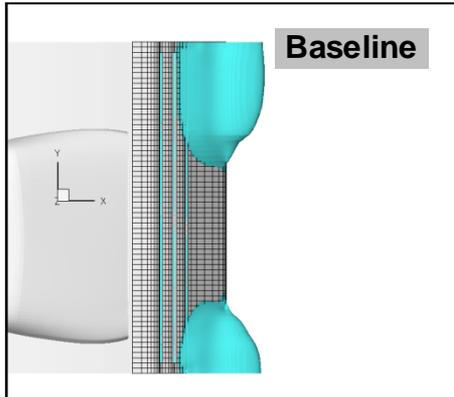
- Continuous/swiveling blowing
 - Swiveling nozzle
 - Louver/vanes system



Sprinkler actuation - results

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- Pair of jets
- Oscillating $\pm 30^\circ$ from side to side
- Frequency of 10Hz



Instantaneous flow structure (limit cycle)

Summary

- **Upper surface blowing is an effective tool in lift enhancement**
- **Degradation in USB effectiveness due to**
 - Premature separation
 - Nozzle edge effects
- **Techniques for lift enhancement:**
 - Geometrical modifications:
 - Flap upper surface
 - Larger aspect-ratio nozzle
 - Flap fences
 - Active Flow Control
 - Constant actuation
 - Pulsed actuation for increase efficiency
 - Sprinkler actuation for increased flexibility